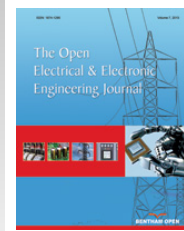




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RESEARCH ARTICLE

Hybrid Active-Passive Robust Fault-Tolerant Control of Event-Triggered Nonlinear NCS

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Abstract: In this paper, the authors aimed to analyze uncertain nonlinear networked control systems (NCS) under discrete event-triggered communication scheme (DETCS), in which an integrated design methodology between robust fault detection observer and active fault-tolerant controller is proposed. Moreover, the problem of hybrid active-passive robust fault-tolerant control, which integrated passive fault-tolerant control, fault detection, and controller reconstruction, is researched. In consideration of the impact of uncertainties and network-induced delay on system performance, a new class of uncertain nonlinear NCS fault model is established based on T-S fuzzy model. By employing Lyapunov stability theory, H_∞ control theory, and linear matrix inequality method, the fault detection observer and hybrid fault-tolerant controller are both appropriately designed. In addition, the sufficient condition that guaranteed the asymptotically robust stability of nonlinear NCS against any actuator failures is deduced. Finally, a numerical simulation is provided to show the effectiveness of the proposed methods.

Keywords: Nonlinear networked control system, Hybrid active-passive robust fault-tolerant control, T-S fuzzy model, Discrete event-triggered communication scheme, Fault detection, PTTCS.

1. INTRODUCTION

The faster development of the modern industry and the complexity in structural design of nonlinear networked control systems (NCS) have led to increased possibility of system faults. A fault-tolerant controller (FTC) is capable of ensuring the stability of the system as well as keeping the performance of the system at an acceptable level. FTC can be achieved either passively by using the off-line designed passive fault-tolerant controller (PFTC) that is robust to predictable failures of the system [1] or actively by using the real-time online fault detection observer (FDO) and controller reconstruction to tolerant unknown faults [2, 3]. However, these two above mentioned methods have disadvantages. A new idea named hybrid active-passive fault-tolerant control, which combined the merits of these two methods, is rarely reported up to now. Most of the researches on FTC for NCS still adopt the period time-triggered communication scheme (PTTCS), whereas the combination between NCS and the discrete event-triggered communication scheme (DETCS), a more efficient way to save network resources, requires further study [4]. The only document that could be referred is [5], which aimed at independent event generator and researched hybrid active-passive fault-tolerant control problem of NCS. Although it has simplified design progress, it still has limitation to the economy of networked resources.

Over the recent years, the model-based approach of fault detection aimed at NCS has received significant attention [6, 7]. However, links between fault detection and FTC are still deficient. Although some recent results on the integration of fault detection and FTC can be found [8, 9], not much are applied to NCS. Therefore, the integration of FDO and hybrid fault-tolerant controller for nonlinear NCS (NNCS) is attracting more and more attention recently. Among all classes of possible faults, actuator fault is considered to be the most easily occurring. On the contrary, most

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literature, which related actuator fault-tolerant control, only addresses linear plants [10]. Most actual systems are nonlinear by nature or at least have nonlinear characteristics [11, 13]. T-S fuzzy model could express nonlinear model as fuzzy fusion of many partial linear models by using fuzzy rules; therefore, it has attracted extensive attention worldwide in the recent years, and many significant problems about it have been deeply researched and has made great progress [14, 17]. Reference [18] only applied passive fault-tolerant techniques to solve the problem of robust guaranteed cost fault-tolerant control for NNCS with network-induced delay and packet dropout. To the author's knowledge, up to now, no report exists about hybrid active-passive robust fault-tolerant control for NNCS based on T-S fuzzy model.

According to the aforementioned analysis, the present paper has researched the controller design problem of hybrid active-passive robust fault-tolerant control for uncertain NNCS against any actuator failure by using appropriately constructed Lyapunov function, H_∞ control theory, and linear matrix inequality (LMI) method, so that the system could realize the following functions: under the premise of saving network resources as much as possible, when known faults appear, PFTC would be tolerant to system faults and maintains stability; when unknown faults appear, PFTC could slow down the rate of system performance deterioration for the sake of FDO that needs time to detect faults; once accurate fault information is detected, the controller will be reconstructed immediately so that the influence of faults could be compensated, and the system could resiliently be stable as soon as possible.

2. MATERIAL AND METHODS

2.1. The Description of NNCS Model

Consider an uncertain NNCS described by the T-S fuzzy model, where the *i*th rule of it is expressed as follows:

$$R^i: \text{if } \theta_1(t) \text{ is } N_{i1} \text{ and } \theta_g(t) \text{ is } N_{ig} \text{ then } \begin{cases} x(t) = (A_i + \Delta A_i)x(t) + (B_i + \Delta B_i)u(t) + E_i f(t), i = 1, 2, \dots, r \\ y(t) = C_i x(t) \end{cases}$$

where *r* is the number of rules; $x(t) \in R^n$ is the state vector; $u(t) \in R^m$, $y(t) \in R^p$ represents the input and output vector of the system, respectively; $N_{is} = (s = 1, 2, \dots, g)$ is the fuzzy set; $\theta(t) = [\theta_1(t), \theta_2(t), \dots, \theta_g(t)]^T$ denotes the premise variables and assumes that it does not rely on $u(t)$; and $f(t) \in R^r$ is the signal of actuator faults, assuming $f(t)$ is an additional fault that satisfied $\|f(t)\| \leq f_0$. When no fault happens, the value is 0, and it will change into a time-varying form or a definite value when faults appear. $A_i, B_i, C_i, E_i (i = 1, 2, \dots, r)$ are constant matrices with appropriate dimensions; $\Delta A_i, \Delta B_i$ denote the uncertainty of system parameter and assumed as norm-bounded. They are time-varying, and described as

$$[\Delta A_i \quad \Delta B_i] = MF(t)[N_{ai} \quad N_{bi}] \tag{1}$$

where M, N_{ai} and N_{bi} are constant matrices with suitable dimensions, respectively; $F(t)$ is an unknown time-varying function matrix whose elements are *Lebesgue* measurable and satisfies $F^T(t)F(t) \leq 1$.

The following T-S fuzzy system state equation can be inferred by using center-average defuzzifier, product inference, and singleton fuzzifier:

$$\begin{cases} x(t) = \sum_{i=1}^r h_i(\theta(t))((A_i + \Delta A_i)x(t) + (B_i + \Delta B_i)u(t) + E_i f(t)) \\ y(t) = \sum_{i=1}^r h_i(\theta(t))C_i x(t) \end{cases} \tag{2}$$

where $h_i(\theta(t)) = \mu_i(\theta(t)) / \sum_{i=1}^r \mu_i(\theta(t))$, $\mu_i(\theta(t)) = \prod_{s=1}^g N_{is}(\theta_s(t))$, and $N_{is}(\theta_s(t))$ are the membership value of

$(\theta_s(t))$ in N_{is} . Suppose $\mu_i(\theta(t)) \geq 0 (i = 1, 2, \dots, r)$ and $\sum_{i=1}^r \mu_i(\theta(t)) > 0$, then $h_i(\theta(t)) \geq 0$ as well as $\sum_{i=1}^r h_i(\theta(t)) = 1$.

2.2. The Introduction of DETCS

For NNCS, DETCS means an introduction of communication constraint condition. Whether or not the sampled state

should be transmitted is determined by the current-sampled state and the state error in sampled period. Only when the event-triggered condition is satisfied, then the data will be transmitted, and vice versa. This will make the system save more network resources under the circumstances of keeping the corresponding expected performance, and therefore increase the efficiency of network utilization.

The structure of the hybrid active-passive fault-tolerant control of NNCS under DETCS is shown in Fig. (1).

For case of convenient exposition, the following assumptions are made:

Assumption 1. Sensors are clock-driven, and controllers and actuators are event-driven. The system states are sampled at a constant period; h, i_k, h is represented for sampled instants; and the set of sampled instants is represented by $\{i_k h | i_k \in N\}$.

Assumption 2. The transmitted instants $t_k h$ are determined by the sampled states $x(i_k h)$. The set of transmission instants is represented by $\{i_k h | i_k \in N\}$, which is a subset of $\{i_k h | i_k \in N\}$.

Assumption 3. The state of the system is completely measured, and it adopts state-feedback control strategy. The delays at transmitted instants $t_k h$ are lumped together as $\tau_{t_k} = \tau_{t_k}^{sc} + \tau_{t_k}^{ca} + \tau_{t_k}^c$, where, $\tau_{t_k}^{sc}$, $\tau_{t_k}^{ca}$ and $\tau_{t_k}^c$ represent the delays from the sensor to the controller, from the controller to the actuator, and computational delays, respectively.

Assumption 4. The detection of fault information is beyond the influence of event-triggered condition, that is, fault information can be transmitted to fault detection observer at every sampled instant.

Assumption 5. The role of the zero order holder (ZOH) is to store the latest data packet. Thus, the actuator keeps the control input unchanged until the output of the ZOH is updated to a new value.

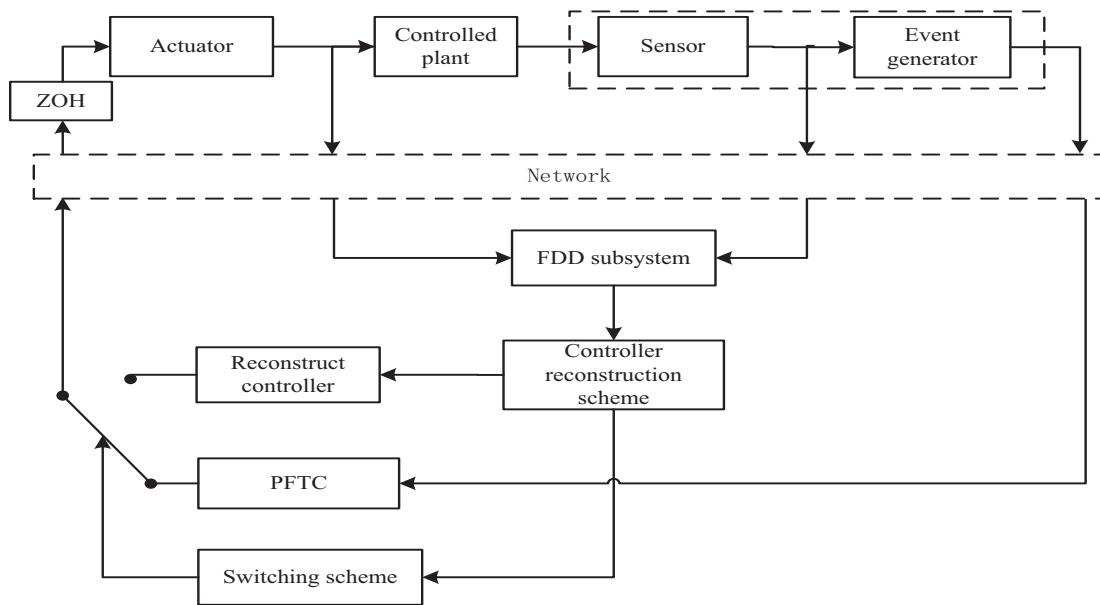


Fig. (1). Structure of the system.

Compared with traditional NCS, the sample data need to pass the event generator before being transmitted by the network, as depicted in Fig. (1), and its function is to determine whether or not to transmit the latest sample signal to the controller. We adopt the following event-triggered condition [19]:

$$t_{k+1}h = t_k h + \min\{lh | e_x^T(i_k h)\Xi e_x(i_k h) \geq \sigma x^T(i_k h)\Xi x(i_k h)\} \tag{3}$$

where

$$e_x(i_k h) = x(i_k h) - x(t_k h) \tag{4}$$

h is the sampling period, Ξ is a symmetric positive definite matrix, and σ is the bounded positive scalar.

Not all the sampled data could be transmitted; only when $x(i_k h)$ and $x(t_k h)$ satisfied the event-triggered condition (3) will the event generator be triggered and data $x(i_k h)$ be transmitted. If the event-triggered condition is designed properly, then the use of DETCS will help the system save more network resources as well as increase the efficiency of resource utilization on the basis of ensuring the system performance at an acceptable level.

2.3. NNCS Fault Model Based on DETCS

From Assumption 3, we can get the truth that the state of the system is completely measurable. Based on the previous description of the controlled plant (2), the feedback controller is designed and the i th rule of the state-feedback control can be described as

$$R^i: \text{ If } \theta_1(t) \text{ is Nil and } \dots \theta_g(t) \dots \text{ is, } N_{ig} \text{ then } u(t) = K_i x(t_k h), t \in [t_k h + \tau_{t_k}, t_{k+1} h + \tau_{t_{k+1}})$$

where K_i ($i = 1, 2, \dots, r$) represent controller gain matrices to be determined in the following theorem; and are network-induced delay at the transmitting instant $t_k h$ and $t_{k+1} h$, respectively.

Meanwhile, considering the role of ZOH when, $t \in [t_k h + \tau_{t_k}, t_{k+1} h + \tau_{t_{k+1}})$ the state-feedback control law, according to the PDC algorithm, is shown as follows:

$$u(t) = \sum_{i=1}^r h_i(\theta(t)) K_i x(t_k h) \tag{5}$$

Define

$$\tau(t) = t - i_k h \tag{6}$$

Obviously, $\tau(t)$ is a continuous linear function that meets the requirement of $0 < \tau_m \leq \tau(t) \leq \tau_M$; where τ_m and τ_M are the lower and upper bounds of network-induced delay, respectively. Define $\tau_s = \tau_M - \tau_m$.

When (4), (5), and (6) are combined, $u(t)$ can be expressed as follows:

$$u(t) = \sum_{i=1}^r h_i(\theta(t)) K_i (x(t - \tau(t)) - e_x(i_k h)) \tag{7}$$

Remark 1: We could see from formula (7) that the control variable contained not only the state variable but also the state error e ($i_k h$) of the event-triggered condition. This makes the computation of the control variable rely on both system state and system error, that is, the introduction of the event-triggered condition.

Supposing fault distribution matrices $E_i = -(B_i + \Delta B_i)$ and, $i = 1, 2, \dots, r$, $Lu(t) = t(t) - f(t)$ for the sake of $f(t)$ is actuator fault, where unknown matrix $L = \text{diag}\{l_1, \dots, l_m\}$, $l_q \in [0, 1]$, $q = 1, 2, \dots, m$ describes the fault extent of system actuators; that is $l_q = 0$, indicates that the q th actuator is totally invalid; $l_q = 1$ indicates that the actuator operates properly; and $l_q \in (0, 1)$ indicates that the q th actuator is at fault to some extent.

The fault model of NNCS can be transformed from (2) to (8) and (9):

$$\begin{cases} x(t) = \sum_{i=1}^r h_i(\theta(t)) ((A_i + \Delta A_i)x(t) + (B_i + \Delta B_i)u(t) - (B_i + \Delta B_i)f(t)) \\ y(t) = \sum_{i=1}^r h_i(\theta(t)) C_i x(t) \end{cases} \tag{8}$$

$$\begin{cases} x(t) = \sum_{i=1}^r h_i(\theta(t)) ((A_i + \Delta A_i)x(t) + (B_i + \Delta B_i)Lu(t)) \\ y(t) = \sum_{i=1}^r h_i(\theta(t)) C_i x(t) \end{cases} \tag{9}$$

NNCS fault model (8) is equivalent to (9), that is, although they are different in form, they have essentially the same meaning. The model integrates event-triggered condition, network-induced delay, model uncertainties, actuator failures,

and control law into a unified framework, which lays a solid foundation for the design of hybrid active-passive fault-tolerant controller for NNCS.

2.4. Related Lemma

Lemma 1 (Schur Complement). For a given symmetric matrix $Z = \begin{bmatrix} Z_{11} & Z_{12} \\ Z_{12}^T & Z_{22} \end{bmatrix}$, the following three conditions are equivalent:

- 1) $Z < 0$;
- 2) $Z_{11} < 0, Z_{22} - Z_{12}^T Z_{11}^{-1} Z_{12} < 0$;
- 3) $Z_{22} < 0, Z_{11} - Z_{12} Z_{22}^{-1} Z_{12}^T < 0$.

Lemma 2. For any positive definite symmetric matrix $W \in R^{n \times n}, W = W^T > 0$, scalar, $0 \leq h(t) \leq h_M$ and vector function $\dot{x} : [-h_M, 0] \rightarrow R^n$, such that the following integration is well defined:

$$-h(t) \int_{-h(t)}^0 x^T(t+s) W x(t+s) ds \leq [x^T(t) \quad x^T(t-h(t))] \begin{bmatrix} -W & W \\ W & -W \end{bmatrix} \begin{bmatrix} x(t) \\ x(t-h(t)) \end{bmatrix}$$

Lemma 3 [20]. Let $f_1, f_2, \dots, f_N : R^m \rightarrow R$ have positive values in an open subset D of R^m . Then, the reciprocally convex combination of f_i over D satisfies

$$\min_{\{\alpha_i | \alpha_i > 0, \sum_i \alpha_i = 1\}} \sum_i \frac{1}{\alpha_i} f_i(t) = \sum_i f_i(t) + \max_{g_{i,j}(t)} \sum_{i \neq j} g_{i,j}(t)$$

Subject to

$$\{g_{i,j} : R^m \rightarrow R, g_{j,i}(t) = g_{i,j}(t), \begin{bmatrix} f_i(t) & g_{i,j}(t) \\ g_{i,j}(t) & f_j(t) \end{bmatrix} \geq 0\}$$

Lemma 4 [21]. Given matrices Y, M, E with appropriate dimensions and $Y = Y^T$, then $Y + MF(t)E + E^T F^T(t)M^T < 0, \forall F : F^T F \leq I$ holds if and only if, for some scalar $\epsilon > 0$ and $Y + \epsilon MM^T + \epsilon^{-1} E^T E < 0$.

3. RESULTS AND DISCUSSION

3.1. Design of the Fault Detection Observer

Considering system model (2) for NNCS, if there is no actuator failure, the system state is observable. Supposing $\hat{x}(t) \in R^n$ represents the estimated state of $x(t)$, then $\hat{y}(t)$ is the output of the observer. According to PDC algorithm and formula (8), fault detection observer is designed as follows:

$$\begin{cases} \hat{\dot{x}}(t) = \sum_{i=1}^r h_i(\theta(t)) ((A_i + \Delta A_i) \hat{x}(t) + (B_i + \Delta B_i) u(t) + G_i (y(t - \tau(t)) - \hat{y}(t - \tau(t)))) \\ \hat{y}(t) = \sum_{i=1}^r h_i(\theta(t)) C_i \hat{x}(t) \end{cases} \tag{10}$$

where G_i is the gain matrix of the state observer.

Residual, state estimate error, and residual error are defined as follows, respectively:

$$r(t) = W(y(t) - \hat{y}(t)) \tag{11}$$

$$e(t) = x(t) - \hat{x}(t) \tag{12}$$

$$r_e(t) = r(t) - f(t) \tag{13}$$

where W is the residual gain matrix and as a result,

$$r(t) = \sum_{i=1}^r h_i(\theta(t)) WC_i e(t) \tag{14}$$

$$e(t) = \sum_{i=1}^r \sum_{j=1}^r h_i(\theta(t)) h_j(\theta(t)) ((A_i + \Delta A_i) e(t) - G_i C_j e(t - \tau(t)) - (B_i + \Delta B_i) f(t)) \tag{15}$$

$$r_e(t) = \sum_{i=1}^r h_i(\theta(t)) (WC_i e(t) - f(t)) \tag{16}$$

The observer-based fault estimation method is derived from H_∞ control theory. In other words, the observer and residual error of real system satisfied such relationship $\|r_e(t)\|_2 \leq \gamma^2 \|f(t)\|_2, f(t) \in L_2[0, \infty)$, where γ is a given constant.

Define H_∞ performance index as $J_1 = \int_0^t (r_e^T(t) r_e(t) - \gamma^2 f^T(t) f(t)) dt$.

Theorem 1. For the given positive scalar $\tau_m, \tau_M, \tau_s, a, b, c, \varepsilon$, if there exists positive definite symmetric matrices, P, V_p, W, Q_i ($i = 1, 2, \dots, 5$) which satisfied the following LMI:

$$\Phi = \begin{bmatrix} \Phi_{11} & \Phi_{12} & \Phi_{13} & \Phi_{14} & \Phi_{15} & \Phi_{16} & \Phi_{17} & \Phi_{18} & \Phi_{19} & \Phi_{110} & \Phi_{111} \\ * & \Phi_{22} & \Phi_{23} & \Phi_{24} & 0 & 0 & \Phi_{27} & \Phi_{28} & \Phi_{29} & 0 & 0 \\ * & * & \Phi_{33} & \Phi_{34} & 0 & 0 & 0 & 0 & 0 & 0 & 0 \\ * & * & * & \Phi_{44} & 0 & 0 & 0 & 0 & 0 & 0 & 0 \\ * & * & * & * & \Phi_{55} & \Phi_{56} & \Phi_{57} & \Phi_{58} & \Phi_{59} & 0 & \Phi_{511} \\ * & * & * & * & * & \Phi_{66} & 0 & 0 & 0 & 0 & 0 \\ * & * & * & * & * & * & \Phi_{77} & 0 & 0 & \Phi_{710} & 0 \\ * & * & * & * & * & * & * & \Phi_{88} & 0 & \Phi_{810} & 0 \\ * & * & * & * & * & * & * & * & \Phi_{99} & \Phi_{910} & 0 \\ * & * & * & * & * & * & * & * & * & \Phi_{1010} & 0 \\ * & * & * & * & * & * & * & * & * & * & \Phi_{1111} \end{bmatrix} \leq 0$$

where * represents the corresponding matrix to be obtained through the symmetric matrix.

$$\begin{aligned} \Phi_{11} &= PA_i + A_i^T P + Q_1 - Q_3 - Q_4, \Phi_{12} = -V_i C_j, \Phi_{13} = Q_3, \Phi_{14} = Q_4, \Phi_{15} = -PB_i, \Phi_{16} = C_j^T W^T, \Phi_{17} = \tau_m A_i^T P, \\ \Phi_{18} &= \tau_M A_i^T P, \Phi_{19} = \tau_s A_i^T P, \Phi_{110} = PM, \Phi_{111} = N_{ai}^T, \Phi_{22} = -2Q_5 + M_{12}^T + M_{12}, \Phi_{23} = Q_5 - M_{12}, \Phi_{24} = Q_5 - M_{12}^T, \\ \Phi_{27} &= -\tau_m C_j^T V_i^T, \Phi_{28} = -\tau_M C_j^T V_i^T, \Phi_{29} = -\tau_s C_j^T V_i^T, \Phi_{33} = Q_2 - Q_1 - Q_3 - Q_5, \Phi_{34} = M_{12}^T, \Phi_{44} = -Q_2 - Q_4 - Q_5, \\ \Phi_{55} &= -\gamma^2 I, \Phi_{56} = -I, \Phi_{57} = -\tau_m B_i^T P, \Phi_{58} = -\tau_M B_i^T P, \Phi_{59} = -\tau_s B_i^T P, \Phi_{511} = -N_{bi}^T, \Phi_{66} = -I, \\ \Phi_{77} &= -2aP + a^2 Q_3, \Phi_{710} = \tau_m PM, \Phi_{88} = -2bP + b^2 Q_4, \Phi_{810} = \tau_M PM, \Phi_{99} = -2cP + c^2 Q_5, \Phi_{910} = \tau_s PM, \\ \Phi_{1010} &= -\varepsilon^{-1} I, \Phi_{1111} = -\varepsilon I \end{aligned}$$

We obtain the observer gain matrix through $G_i = P^{-1} V_i$ and the fault detection observer could ensure the fault

estimation error meets the requirement of $\|r_e(t)\|_2 \leq \gamma^2 \|f(t)\|_2$.

Proof.

Constructing Lyapunov–Krasovskii function is shown as follows:

$$V(t) = e^T(t)Pe(t) + \int_{t-\tau_m}^t e^T(s)Q_1e(s)ds + \int_{t-\tau_M}^{t-\tau_m} e^T(s)Q_2e(s)ds + \int_{-\tau_m}^0 \int_{t+\theta}^t \tau_m \dot{e}^T(s)Q_3\dot{e}(s)dsd\theta + \int_{-\tau_M}^0 \int_{t+\theta}^t \tau_M \dot{e}^T(s)Q_4\dot{e}(s)dsd\theta + \int_{-\tau_M}^{-\tau_m} \int_{t+\theta}^t \tau_s \dot{e}^T(s)Q_5\dot{e}(s)dsd\theta$$

For the sake of convenience in writing, we take the following abbreviated forms:

$$e(t) = e, e(t - \tau(t)) = e_\tau, e(t - \tau_m) = e_{\tau_m}, e(t - \tau_M) = e_{\tau_M}, h_i(\theta(t)) = h_i, h_j(\theta(t)) = h_j$$

Taking the derivation of $V(t)$ along the trajectory of (15), we obtain

$$\dot{V}(t) = 2e^T P\dot{e} + e^T Q_1\dot{e} - e_{\tau_m}^T Q_1\dot{e}_{\tau_m} + e_{\tau_m}^T Q_2\dot{e}_{\tau_m} - e_{\tau_M}^T Q_2\dot{e}_{\tau_M} + \tau_m^2 \dot{e}^T Q_3\dot{e} - \tau_m \int_{t-\tau_m}^t \dot{e}^T Q_3\dot{e}ds + \tau_M^2 \dot{e}^T Q_4\dot{e} - \tau_M \int_{t-\tau_M}^t \dot{e}^T Q_4\dot{e}ds + \tau_s^2 \dot{e}^T Q_5\dot{e} - \tau_s \int_{t-\tau_M}^{t-\tau_m} \dot{e}^T Q_5\dot{e}ds \tag{17}$$

According to lemma 2, we have

$$-\tau_m \int_{t-\tau_m}^t e^T Q_3 \dot{e} ds \leq \begin{bmatrix} e \\ e_{\tau_m} \end{bmatrix}^T \begin{bmatrix} -Q_3 & Q_3 \\ * & -Q_3 \end{bmatrix} \begin{bmatrix} e \\ e_{\tau_m} \end{bmatrix} \tag{18}$$

$$-\tau_M \int_{t-\tau_M}^t e^T Q_4 \dot{e} ds \leq \begin{bmatrix} e \\ e_{\tau_M} \end{bmatrix}^T \begin{bmatrix} -Q_4 & Q_4 \\ * & -Q_4 \end{bmatrix} \begin{bmatrix} e \\ e_{\tau_M} \end{bmatrix} \tag{19}$$

According to lemma 3, we obtain

$$-\tau_s \int_{t-\tau_M}^{t-\tau_m} e^T Q_5 \dot{e} ds \leq - \begin{bmatrix} e_\tau - e_{\tau_M} \\ e_{\tau_m} - e_\tau \end{bmatrix}^T \begin{bmatrix} Q_5 & M_{12} \\ * & Q_5 \end{bmatrix} \begin{bmatrix} e_\tau - e_{\tau_M} \\ e_{\tau_m} - e_\tau \end{bmatrix} \text{ where } \begin{bmatrix} Q_5 & M_{12} \\ * & Q_5 \end{bmatrix} \geq 0 \tag{20}$$

$$\begin{cases} \tau_m^2 \dot{e}^T Q_3 \dot{e} = \tau_m^2 \xi^T(t) \varepsilon'^T Q_3 \varepsilon' \xi(t) \\ \tau_M^2 \dot{e}^T Q_4 \dot{e} = \tau_M^2 \xi^T(t) \varepsilon'^T Q_4 \varepsilon' \xi(t) \\ \tau_s^2 \dot{e}^T Q_5 \dot{e} = \tau_s^2 \xi^T(t) \varepsilon'^T Q_5 \varepsilon' \xi(t) \end{cases} \tag{21}$$

where,

$$\varepsilon' = \begin{bmatrix} (A_i + \Delta A_i) & -G_i C_j & 0 & 0 & -(B_i + \Delta B_i) \end{bmatrix}, \xi^T(t) = \begin{bmatrix} e^T & e_\tau^T & e_{\tau_m}^T & e_{\tau_M}^T & f^T(t) \end{bmatrix}$$

When (17)-(21) are combined, we have $V(t) - \gamma^2 f^T(t) f(t) \leq \sum_{i=1}^r \sum_{j=1}^r h_i h_j (\xi^T(t) (\Phi' + \tau_m^2 \varepsilon'^T Q_3 \varepsilon' + \tau_M^2 \varepsilon'^T Q_4 \varepsilon' + \tau_s^2 \varepsilon'^T Q_5 \varepsilon') \xi(t))$,

$$\Phi' = \begin{bmatrix} \Phi'_{11} & \Phi'_{12} & \Phi'_{13} & \Phi'_{14} & \Phi'_{15} \\ * & \Phi'_{22} & \Phi'_{23} & \Phi'_{24} & 0 \\ * & * & \Phi'_{33} & \Phi'_{34} & 0 \\ * & * & * & \Phi'_{44} & 0 \\ * & * & * & * & \Phi'_{55} \end{bmatrix} < 0$$

where * represents the corresponding matrix to be obtained through the symmetric matrix.

$$\begin{aligned} \Phi'_{11} &= P(A_i + \Delta A_i) + (A_i + \Delta A_i)^T P + Q_1 - Q_3 - Q_4, \Phi'_{12} = -PG_i C_j, \Phi'_{13} = Q_3, \Phi'_{14} = Q_4, \Phi'_{15} = -P(B_i + \Delta B_i), \\ \Phi'_{22} &= -2Q_5 + M_{12}^T + M_{12} \Phi'_{23} = Q_5 - M_{12}, \Phi'_{24} = Q_5 - M_{12}^T, \Phi'_{33} = Q_2 - Q_1 - Q_3 - Q_5, \Phi'_{34} = M_{12}^T, \\ \Phi'_{44} &= -Q_2 - Q_4 - Q_5, \Phi'_{55} = -\gamma^2 I \end{aligned}$$

From formula (16), we have $r_e^T(t)r_e(t) = \xi^T(t)\Sigma^T\Sigma\xi(t)$, $\Sigma = [WC_i \ 0 \ 0 \ 0 \ -I]$.

Therefore, from the above derivation, we obtain

$$V(t) + \dot{r}_e^T(t)r_e(t) - \gamma^2 f^T(t)f(t) \leq \sum_{i=1}^r \sum_{j=1}^r h_i h_j \left(\xi^T(t) (\Phi' + \tau_m^2 \varepsilon'^T Q_3 \varepsilon' + \tau_M^2 \varepsilon'^T Q_4 \varepsilon' + \tau_s^2 \varepsilon'^T Q_5 \varepsilon' + \Sigma^T \Sigma) \xi(t) \right)$$

Set $\Phi'' = \Phi' + \tau_m^2 \varepsilon'^T Q_3 \varepsilon' + \tau_M^2 \varepsilon'^T Q_4 \varepsilon' + \tau_s^2 \varepsilon'^T Q_5 \varepsilon' + \Sigma^T \Sigma$, and applying Schur Complement to Φ'' , we obtain

$$\Phi'' = \begin{bmatrix} \Phi''_{11} & \Phi''_{12} & \Phi''_{13} & \Phi''_{14} & \Phi''_{15} & \Phi''_{16} & \Phi''_{17} & \Phi''_{18} & \Phi''_{19} \\ * & \Phi''_{22} & \Phi''_{23} & \Phi''_{24} & 0 & 0 & \Phi''_{27} & \Phi''_{28} & \Phi''_{29} \\ * & * & \Phi''_{33} & \Phi''_{34} & 0 & 0 & 0 & 0 & 0 \\ * & * & * & \Phi''_{44} & 0 & 0 & 0 & 0 & 0 \\ * & * & * & * & \Phi''_{55} & \Phi''_{56} & \Phi''_{57} & \Phi''_{58} & \Phi''_{59} \\ * & * & * & * & * & \Phi''_{66} & 0 & 0 & 0 \\ * & * & * & * & * & * & \Phi''_{77} & 0 & 0 \\ * & * & * & * & * & * & * & \Phi''_{88} & 0 \\ * & * & * & * & * & * & * & * & \Phi''_{99} \end{bmatrix} < 0$$

where * represents the corresponding matrix to be obtained through the symmetric matrix.

$$\begin{aligned} \Phi''_{11} &= P(A_i + \Delta A_i) + (A_i + \Delta A_i)^T P + Q_1 - Q_3 - Q_4, \Phi''_{12} = -PG_i C_j, \Phi''_{13} = Q_3, \Phi''_{14} = Q_4, \Phi''_{15} = -P(B_i + \Delta B_i), \\ \Phi''_{16} &= C_j^T W^T, \Phi''_{17} = \tau_m (A_i + \Delta A_i)^T Q_3, \Phi''_{18} = \tau_M (A_i + \Delta A_i)^T Q_4, \Phi''_{19} = \tau_s (A_i + \Delta A_i)^T Q_5, \\ \Phi''_{22} &= -2Q_5 + M_{12}^T + M_{12}, \Phi''_{23} = Q_5 - M_{12}, \Phi''_{24} = Q_5 - M_{12}^T, \Phi''_{27} = -\tau_m C_j^T G_i^T Q_3, \Phi''_{28} = -\tau_M C_j^T G_i^T Q_4, \\ \Phi''_{29} &= -\tau_s C_j^T G_i^T Q_5, \Phi''_{33} = Q_2 - Q_1 - Q_3 - Q_5, \Phi''_{34} = M_{12}^T, \Phi''_{44} = -Q_2 - Q_4 - Q_5, \Phi''_{55} = -\gamma^2 I, \Phi''_{56} = -I, \\ \Phi''_{57} &= -\tau_m (B_i + \Delta B_i)^T Q_3, \Phi''_{58} = -\tau_M (B_i + \Delta B_i)^T Q_4, \Phi''_{59} = -\tau_s (B_i + \Delta B_i)^T Q_5, \Phi''_{66} = -I, \\ \Phi''_{77} &= -Q_3, \Phi''_{88} = -Q_4, \Phi''_{99} = -Q_5 \end{aligned}$$

According to lemma 4, transforming Φ'' into, $\Phi''' = \Phi'' + \varepsilon M_1 M_1^T + \varepsilon^{-1} E_1^T E_1$ where

$$M_1 = \begin{bmatrix} M^T P & 0 & 0 & 0 & 0 & 0 & \tau_m M^T Q_3 & \tau_M M^T Q_4 & \tau_s M^T Q_5 \end{bmatrix}^T, E_1 = \begin{bmatrix} N_{ai} & 0 & 0 & 0 & -N_{bi} & 0 & 0 & 0 & 0 \end{bmatrix}$$

$$\Phi''' = \begin{bmatrix} \Phi'''_{11} & \Phi'''_{12} & \Phi'''_{13} & \Phi'''_{14} & \Phi'''_{15} & \Phi'''_{16} & \Phi'''_{17} & \Phi'''_{18} & \Phi'''_{19} \\ * & \Phi'''_{22} & \Phi'''_{23} & \Phi'''_{24} & 0 & 0 & \Phi'''_{27} & \Phi'''_{28} & \Phi'''_{29} \\ * & * & \Phi'''_{33} & \Phi'''_{34} & 0 & 0 & 0 & 0 & 0 \\ * & * & * & \Phi'''_{44} & 0 & 0 & 0 & 0 & 0 \\ * & * & * & * & \Phi'''_{55} & \Phi'''_{56} & \Phi'''_{57} & \Phi'''_{58} & \Phi'''_{59} \\ * & * & * & * & * & \Phi'''_{66} & 0 & 0 & 0 \\ * & * & * & * & * & * & \Phi'''_{77} & 0 & 0 \\ * & * & * & * & * & * & * & \Phi'''_{88} & 0 \\ * & * & * & * & * & * & * & * & \Phi'''_{99} \end{bmatrix} \leq 0$$

where * represents the corresponding matrix to be obtained through the symmetric matrix.

$$\begin{aligned} \Phi_{11}'' &= PA_i + A_i^T P + Q_1 - Q_3 - Q_4, \Phi_{12}'' = -PG_i C_j, \Phi_{13}'' = Q_3, \Phi_{14}'' = Q_4, \Phi_{15}'' = -PB_i, \Phi_{16}'' = C_j^T W^T, \Phi_{17}'' = \tau_m A_i^T Q_3, \\ \Phi_{18}'' &= \tau_M A_i^T Q_4, \Phi_{19}'' = \tau_s A_i^T Q_5, \Phi_{22}'' = -2Q_5 + M_{12}^T + M_{12}, \Phi_{23}'' = Q_5 - M_{12}, \Phi_{24}'' = Q_5 - M_{12}^T, \Phi_{27}'' = -\tau_m C_j^T G_i^T Q_3, \\ \Phi_{28}'' &= -\tau_M C_j^T G_i^T Q_4, \Phi_{29}'' = -\tau_s C_j^T G_i^T Q_5, \Phi_{33}'' = Q_2 - Q_1 - Q_3 - Q_5, \Phi_{34}'' = M_{12}^T, \Phi_{44}'' = -Q_2 - Q_4 - Q_5, \Phi_{55}'' = -\gamma^2 I, \\ \Phi_{56}'' &= -I, \Phi_{57}'' = -\tau_m B_i^T Q_3, \Phi_{58}'' = -\tau_M B_i^T Q_4, \Phi_{59}'' = -\tau_s B_i^T Q_5, \Phi_{66}'' = -I, \Phi_{77}'' = -Q_3, \Phi_{88}'' = -Q_4, \Phi_{99}'' = -Q_5 \end{aligned}$$

Applying Schur Complement to Φ'' , we obtain

$$\Phi'' = \begin{bmatrix} \Phi_{11}'' & \Phi_{12}'' & \Phi_{13}'' & \Phi_{14}'' & \Phi_{15}'' & \Phi_{16}'' & \Phi_{17}'' & \Phi_{18}'' & \Phi_{19}'' & PM & N_{ai}^T \\ * & \Phi_{22}'' & \Phi_{23}'' & \Phi_{24}'' & 0 & 0 & \Phi_{27}'' & \Phi_{28}'' & \Phi_{29}'' & 0 & 0 \\ * & * & \Phi_{33}'' & \Phi_{34}'' & 0 & 0 & 0 & 0 & 0 & 0 & 0 \\ * & * & * & \Phi_{44}'' & 0 & 0 & 0 & 0 & 0 & 0 & 0 \\ * & * & * & * & \Phi_{55}'' & \Phi_{56}'' & \Phi_{57}'' & \Phi_{58}'' & \Phi_{59}'' & 0 & -N_{bi}^T \\ * & * & * & * & * & \Phi_{66}'' & 0 & 0 & 0 & 0 & 0 \\ * & * & * & * & * & * & \Phi_{77}'' & 0 & 0 & \tau_m Q_3 M & 0 \\ * & * & * & * & * & * & * & \Phi_{88}'' & 0 & \tau_M Q_4 M & 0 \\ * & * & * & * & * & * & * & * & \Phi_{99}'' & \tau_s Q_5 M & 0 \\ * & * & * & * & * & * & * & * & * & -\varepsilon^{-1} I & 0 \\ * & * & * & * & * & * & * & * & * & * & -\varepsilon I \end{bmatrix} \leq 0$$

where * represents the corresponding matrix to be obtained through the symmetric matrix.

Before and after multiplying Φ'' with $\text{diag} \{I, I, I, I, I, I, PQ_3^{-1}, PQ_4^{-1}, PQ_5^{-1}, I, I\}$ and from the following inequality

$$-a^2 Q_3 + 2aP - PQ_3^{-1} P \leq -(aQ_3 - P)Q_3^{-1}(aQ_3 - P) \leq 0, \text{ we obtain } \begin{cases} -PQ_3^{-1} P \leq -2aP + a^2 Q_3 \\ -PQ_4^{-1} P \leq -2bP + b^2 Q_4 \\ -PQ_5^{-1} P \leq -2cP + c^2 Q_5 \end{cases}$$

Define $PG_i = V_i$, then we obtain Φ .

As previously described, we could make the following conclusions:

$$V(t) + r_e^T(t)r_e(t) - \gamma^2 f^T(t)f(t) \leq \sum_{i=1}^r \sum_{j=1}^r h_i h_j \xi^T(t) \Phi \xi^T(t) \text{ When formula } f(t) = 0 \text{ is true, and LMI } \Phi < 0, \text{ then } \dot{V}(t) \leq 0 \text{ could be driven and the error system is asymptotically stable.}$$

For any non-zero, $f(t) \in [0, \infty)$ the integral previous formula from $t_0 \rightarrow t$ and we obtain

$$V(t) - V(t_0) \leq -\int_{t_0}^t (r_e^T(t)r_e(t) - \gamma^2 f^T(t)f(t)) dt$$

Under zero initial condition, when $t \rightarrow \infty$, $\int_0^\infty r_e^T(t)r_e(t) dt \leq \int_0^\infty \gamma^2 f^T(t)f(t) dt$, is tenable, that is, $\|r_e(t)\|_2 \leq \gamma \|f(t)\|_2$. Thus, the error system (16) finally uniformly bounded as well as satisfied the previous H_∞ performance index; furthermore, the observer gain matrix could be calculated through $G_i = P^{-1}V_i$.

The proof to theorem 1 is hereby completed.

Remark 2: In this proof, we adopted an improved Jensen inequality technique, which possesses tighter integration upper and lower bounds. For T-S fuzzy system, the application of this method is computationally efficient while possessing less conservativeness compared with liberty matrix method.

3.2. Design of Hybrid Fault-tolerant Controller

3.2.1. Design of AFTC

Assuming $\text{rank}(F, E_i) = \text{rank}(F)$ is true first, according to the theorem from literature [22], under the

circumstances of the previous assumptions, there exists a matrix $F^+ \in R^{m \times n}$ that makes the following equality $(I + FF^+)E_i = 0$, workable.

The set hybrid fault-tolerant compensation controller based on fault detection observer is

$$u(t) = u_p(t) - F^+ E_i \hat{f}(t) \tag{22}$$

where F^+ is the right false inverse matrix of $B_i + \Delta B_i$, $u_p(t)$ is the designed PFTC from theorem 3, and $\hat{f}(t)$ represents the estimate value of fault.

For the sake of convenience in writing, we take the following abbreviated forms:

$$e_f(t) = f(t) - \hat{f}(t), \quad \hat{f}(t) = \hat{f}, \quad f(t) = f, \quad x(t - \tau_M) = x_{\tau_M}, \quad x(t) = x, \quad x(t - \tau(t)) = x_\tau, \quad x(t - \tau_m) = x_{\tau_m}, \\ h_i(\theta(t)) = h_i, \quad h_j(\theta(t)) = h_j.$$

Applying (22) to (8), we obtain

$$\begin{cases} x(t) = \sum_{i=1}^r h_i \left((A_i + \Delta A_i)x + (B_i + \Delta B_i)(u_p(t) - F^+ E_i \hat{f}) + E_i f \right) \\ \quad = \sum_{i=1}^r h_i h_j \left((A_i + \Delta A_i)x + (B_i + \Delta B_i)(K_j(x_\tau - e_x(i_k h)) - F^+ E_i \hat{f}) + E_i f \right) \\ \quad = \sum_{i=1}^r h_i h_j \left((A_i + \Delta A_i)x + (B_i + \Delta B_i)K_j x_\tau - (B_i + \Delta B_i)K_j e_x(i_k h) + E_i e_f(t) \right) \\ y(t) = \sum_{i=1}^r h_i C_i x \end{cases} \tag{23}$$

where $e_f(t)$ represents fault estimate error.

In consideration of FDO designed in theorem 1, it could ensure that the error system is asymptotically stable, so $e_f(t)$ can be seen a kind of external disturbance $W(t)$, and formula (23) can be transformed into (24):

$$\begin{cases} x(t) = \sum_{i=1}^r h_i h_j \left((A_i + \Delta A_i)x + (B_i + \Delta B_i)K_j x_\tau - (B_i + \Delta B_i)K_j e_x(i_k h) + D_i w(t) \right) \\ y(t) = \sum_{i=1}^r h_i C_i x \end{cases} \tag{24}$$

where, $D_i = E_i = - (B_i + \Delta B_i)$, $w(t) = e_f(t)$.

Theorem 2. Under the event-triggered condition (3) in the DETCS, for the given constants, $\sigma \in [0, 1)$, $\tau_m, \tau_M, \tau_s, \varepsilon, \gamma$ the hybrid fault-tolerant controller (22) could keep the system stable and satisfy the following H_∞ performance index:

$$J_2 = \int_0^t \left(y^T(t) y(t) - \gamma^2 w^T(t) w(t) \right) dt \tag{25}$$

If there exists some positive definite symmetric matrices $R_i > 0, i = 1, 2, \dots, 5$ and V, Y_j . For any possible actuator failures in mode set L and acceptable parameter uncertainties, these parameters satisfied the following LMI:

$$\Phi = \begin{bmatrix} \Phi_{11} & \Phi_{12} & \Phi_{13} & \Phi_{14} & \Phi_{15} & \Phi_{16} & \Phi_{17} & \Phi_{18} & \Phi_{19} & \Phi_{110} & \Phi_{111} & \Phi_{112} \\ * & \Phi_{22} & \Phi_{23} & \Phi_{24} & 0 & 0 & \Phi_{27} & \Phi_{28} & \Phi_{29} & 0 & \Phi_{211} & 0 \\ * & * & \Phi_{33} & \Phi_{34} & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 \\ * & * & * & \Phi_{44} & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 \\ * & * & * & * & \Phi_{55} & 0 & \Phi_{57} & \Phi_{58} & \Phi_{59} & 0 & \Phi_{511} & 0 \\ * & * & * & * & * & \Phi_{66} & \Phi_{67} & \Phi_{68} & \Phi_{69} & 0 & \Phi_{611} & 0 \\ * & * & * & * & * & * & \Phi_{77} & 0 & 0 & \Phi_{710} & 0 & 0 \\ * & * & * & * & * & * & * & \Phi_{88} & 0 & \Phi_{810} & 0 & 0 \\ * & * & * & * & * & * & * & * & \Phi_{99} & \Phi_{910} & 0 & 0 \\ * & * & * & * & * & * & * & * & * & \Phi_{1010} & 0 & 0 \\ * & * & * & * & * & * & * & * & * & * & \Phi_{1111} & 0 \\ * & * & * & * & * & * & * & * & * & * & * & \Phi_{1212} \end{bmatrix} < 0$$

where * represents the corresponding matrix to be obtained through the symmetric matrix.

$$\begin{aligned} \Phi_{11} &= A_i X + X A_i^T - 2X - R_1 + R_3 + R_4, \Phi_{12} = B_i Y_j, \Phi_{13} = 2X - R_3, \Phi_{14} = 2X - R_4, \Phi_{15} = -B_i Y_j, \Phi_{16} = -B_i X, \\ \Phi_{17} &= \tau_m X A_i^T, \Phi_{18} = \tau_M X A_i^T, \Phi_{19} = \tau_s X A_i^T, \Phi_{110} = M, \Phi_{111} = X N_{ai}^T, \Phi_{112} = C_i X^T, \\ \Phi_{22} &= \sigma(2X - V) + 2R_5 - R_6 - R_6^T, \Phi_{23} = R_6 - R_5, \Phi_{24} = R_6^T - R_5, \Phi_{27} = \tau_m Y_j^T B_i^T, \Phi_{28} = \tau_M Y_j^T B_i^T, \Phi_{29} = \tau_s Y_j^T B_i^T, \\ \Phi_{211} &= Y_j^T N_{bi}^T, \Phi_{33} = R_1 + R_3 + R_5 - R_2 - 4X, \Phi_{34} = 2X - R_6^T, \Phi_{44} = R_2 + R_4 + R_5 - 6X, \Phi_{55} = -2X + V, \\ \Phi_{57} &= -\tau_m Y_j^T B_i^T, \Phi_{58} = -\tau_M Y_j^T B_i^T, \Phi_{59} = -\tau_s Y_j^T B_i^T, \Phi_{511} = -Y_j^T N_{bi}^T, \Phi_{66} = -X \gamma^2 I X, \Phi_{67} = -\tau_m X B_i^T, \\ \Phi_{68} &= -\tau_M X B_i^T, \Phi_{69} = -\tau_s X B_i^T, \Phi_{611} = -X N_{bi}^T, \Phi_{77} = -R_3, \Phi_{710} = \tau_m M, \Phi_{88} = -R_4, \Phi_{810} = \tau_M M, \\ \Phi_{99} &= -R_5, \Phi_{910} = \tau_s M, \Phi_{1010} = -\varepsilon^{-1} I, \Phi_{1111} = -\varepsilon I, \Phi_{1212} = -I \end{aligned}$$

Proof.

Constructing Lyapunov–Krasovskii function as the following:

$$\begin{aligned} V(t) &= x^T(t) P x(t) + \int_{t-\tau_m}^t x^T(s) Q_1 x(s) ds + \int_{t-\tau_M}^{t-\tau_m} x^T(s) Q_2 x(s) ds + \int_{-\tau_m}^0 \int_{t+\theta}^t \tau_m x^T(s) Q_3 x(s) ds d\theta + \\ &\int_{-\tau_M}^0 \int_{t+\theta}^t \tau_M \dot{x}^T(s) Q_4 \dot{x}(s) ds d\theta + \int_{-\tau_m}^{-\tau_M} \int_{t+\theta}^t \tau_s \dot{x}^T(s) Q_5 \dot{x}(s) ds d\theta \end{aligned}$$

where $P^T = P > 0, Q_i^T = Q_i > 0, i = 1, 2, 3, 4, 5$.

Taking the derivation of $V(t)$ along the trajectory of (24), we obtain

$$\begin{aligned} \dot{V}(t) &= 2x^T P \dot{x} + x^T Q_1 \dot{x} - x_{\tau_m}^T Q_1 x_{\tau_m} + x_{\tau_m}^T Q_2 x_{\tau_m} - x_{\tau_M}^T Q_2 x_{\tau_M} + \tau_m^2 \dot{x}^T Q_3 \dot{x} - \tau_m \int_{t-\tau_m}^t \dot{x}^T Q_3 \dot{x} ds + \tau_M^2 \dot{x}^T Q_4 \dot{x} - \\ &\tau_M \int_{t-\tau_M}^t \dot{x}^T Q_4 \dot{x} ds + \tau_s^2 \dot{x}^T Q_5 \dot{x} - \tau_s \int_{t-\tau_M}^{t-\tau_m} \dot{x}^T Q_5 \dot{x} ds + e_x^T(i_k h) \Xi e_x(i_k h) - e_x^T(i_k h) \Xi e_x(i_k h) + y^T(t) y(t) - \\ &\gamma^2 w^T(t) w(t) + \gamma^2 w^T(t) w(t) - y^T(t) y(t) \end{aligned}$$

For transmission mechanism (3), when $i_k h \in [t_k h, t_{k+1} h]$, then the following inequality:

$$e_x^T(i_k h) \Xi e_x(i_k h) \leq \sigma x^T(i_k h) \Xi x(i_k h) \text{ is true.}$$

Meanwhile, according to lemma 2, we have

$$-\tau_m \int_{t-\tau_m}^t x^T Q_3 x ds \leq \begin{bmatrix} x \\ x_{\tau_m} \end{bmatrix}^T \begin{bmatrix} -Q_3 & Q_3 \\ * & -Q_3 \end{bmatrix} \begin{bmatrix} x \\ x_{\tau_m} \end{bmatrix}, -\tau_M \int_{t-\tau_M}^t x^T Q_4 x ds \leq \begin{bmatrix} x \\ x_{\tau_M} \end{bmatrix}^T \begin{bmatrix} -Q_4 & Q_4 \\ * & -Q_4 \end{bmatrix} \begin{bmatrix} x \\ x_{\tau_M} \end{bmatrix}$$

According to lemma 3, we obtain

$$-\tau_s \int_{t-\tau_M}^{t-\tau_m} x^T Q_5 \dot{x} ds \leq - \begin{bmatrix} x_\tau - x_{\tau_M} \\ x_{\tau_m} - x_\tau \end{bmatrix}^T \begin{bmatrix} Q_5 & M_{12} \\ * & Q_5 \end{bmatrix} \begin{bmatrix} x_\tau - x_{\tau_M} \\ x_{\tau_m} - x_\tau \end{bmatrix} \text{ where } \begin{bmatrix} Q_5 & M_{12} \\ * & Q_5 \end{bmatrix} \geq 0$$

$$\begin{cases} \tau_m^2 \dot{x}^T Q_3 \dot{x} = \tau_m^2 \eta^T(t) \varepsilon'^T Q_3 \varepsilon' \eta(t) \\ \tau_M^2 \dot{x}^T Q_4 \dot{x} = \tau_M^2 \eta^T(t) \varepsilon'^T Q_4 \varepsilon' \eta(t) \\ \tau_s^2 \dot{x}^T Q_5 \dot{x} = \tau_s^2 \eta^T(t) \varepsilon'^T Q_5 \varepsilon' \eta(t) \end{cases}$$

where $\varepsilon' = [(A_i + \Delta A_i) \quad (B_i + \Delta B_i)K_j \quad 0 \quad 0 \quad -(B_i + \Delta B_i)K_j \quad -(B_i + \Delta B_i)]$, $\eta^T(t) = [x^T \quad x_\tau^T \quad x_{\tau_m}^T \quad x_{\tau_M}^T \quad e_x^T(i, h) \quad w^T(t)]$

The proof process is same as *theorem 1*, and then we have:

$$V + y^T(t)y(t) - \gamma^2 w^T(t)w(t) \leq \sum_{i=1}^r \sum_{j=1}^r h_i h_j \left(\eta^T(t) \left(\Phi' + \tau_m^2 \varepsilon'^T Q_3 \varepsilon' + \tau_M^2 \varepsilon'^T Q_4 \varepsilon' + \tau_s^2 \varepsilon'^T Q_5 \varepsilon' \right) \eta(t) \right)$$

$$\Phi' = \begin{bmatrix} \Phi'_{11} & \Phi'_{12} & \Phi'_{13} & \Phi'_{14} & \Phi'_{15} & \Phi'_{16} \\ * & \Phi'_{22} & \Phi'_{23} & \Phi'_{24} & 0 & 0 \\ * & * & \Phi'_{33} & \Phi'_{34} & 0 & 0 \\ * & * & * & \Phi'_{44} & 0 & 0 \\ * & * & * & * & \Phi'_{55} & 0 \\ * & * & * & * & * & \Phi'_{66} \end{bmatrix} < 0$$

where * represents the corresponding matrix to be obtained through the symmetric matrix.

$$\begin{aligned} \Phi'_{11} &= P(A_i + \Delta A_i) + (A_i + \Delta A_i)^T P + Q_1 - Q_3 - Q_4 + C_i^T C_i, \quad \Phi'_{12} = P(B_i + \Delta B_i)K_j, \quad \Phi'_{13} = Q_3, \Phi'_{14} = Q_4, \Phi'_{15} = -P(B_i + \Delta B_i)K_j, \\ \Phi'_{16} &= -P(B_i + \Delta B_i), \Phi'_{22} = \sigma \Xi - 2Q_5 + M_{12}^T + M_{12}, \Phi'_{23} = Q_5 - M_{12}, \quad \Phi'_{24} = Q_5 - M_{12}^T, \Phi'_{33} = Q_2 - Q_1 - Q_3 - Q_5, \\ \Phi'_{34} &= M_{12}^T, \Phi'_{44} = -Q_2 - Q_4 - Q_5, \quad \Phi'_{55} = -\Xi, \Phi'_{66} = -\gamma^2 I \end{aligned}$$

Define $\Phi'' = \Phi' + \tau_m^2 \varepsilon'^T Q_3 \varepsilon' + \tau_M^2 \varepsilon'^T Q_4 \varepsilon' + \tau_s^2 \varepsilon'^T Q_5 \varepsilon'$

Applying Schur Complement to Φ'' , we have

$$\Phi'' = \begin{bmatrix} \Phi''_{11} & \Phi''_{12} & \Phi''_{13} & \Phi''_{14} & \Phi''_{15} & \Phi''_{16} & \Phi''_{17} & \Phi''_{18} & \Phi''_{19} \\ * & \Phi''_{22} & \Phi''_{23} & \Phi''_{24} & 0 & 0 & \Phi''_{27} & \Phi''_{28} & \Phi''_{29} \\ * & * & \Phi''_{33} & \Phi''_{34} & 0 & 0 & 0 & 0 & 0 \\ * & * & * & \Phi''_{44} & 0 & 0 & 0 & 0 & 0 \\ * & * & * & * & \Phi''_{55} & 0 & \Phi''_{57} & \Phi''_{58} & \Phi''_{59} \\ * & * & * & * & * & \Phi''_{66} & \Phi''_{67} & \Phi''_{68} & \Phi''_{69} \\ * & * & * & * & * & * & \Phi''_{77} & 0 & 0 \\ * & * & * & * & * & * & * & \Phi''_{88} & 0 \\ * & * & * & * & * & * & * & * & \Phi''_{99} \end{bmatrix} < 0$$

where * represents the corresponding matrix to be obtained through the symmetric matrix.

$$\begin{aligned} \Phi_{11}'' &= P(A_i + \Delta A_i) + (A_i + \Delta A_i)^T P + Q_1 - Q_3 - Q_4 + C_i^T C_i \Phi_{12}'' = P(B_i + \Delta B_i) K_j, \Phi_{13}'' = Q_3, \Phi_{14}'' = Q_4, \Phi_{15}'' = -P(B_i + \Delta B_i) K_j, \\ \Phi_{16}'' &= -P(B_i + \Delta B_i), \Phi_{17}'' = \tau_m (A_i + \Delta A_i)^T Q_3, \Phi_{18}'' = \tau_m (A_i + \Delta A_i)^T Q_4, \Phi_{19}'' = \tau_s (A_i + \Delta A_i)^T Q_5, \\ \Phi_{22}'' &= \sigma \Xi - 2Q_5 + M_{12}^T + M_{12}, \Phi_{23}'' = Q_5 - M_{12}, \Phi_{24}'' = Q_5 - M_{12}^T, \Phi_{27}'' = \tau_m K_j^T (B_i + \Delta B_i)^T Q_3, \\ \Phi_{28}'' &= \tau_m K_j^T (B_i + \Delta B_i)^T Q_4, \Phi_{29}'' = \tau_s K_j^T (B_i + \Delta B_i)^T Q_5, \Phi_{33}'' = Q_2 - Q_1 - Q_3 - Q_5, \Phi_{34}'' = M_{12}^T, \\ \Phi_{44}'' &= -Q_2 - Q_4 - Q_5, \Phi_{55}'' = -\Xi, \Phi_{57}'' = -\tau_m K_j^T (B_i + \Delta B_i)^T Q_3, \Phi_{58}'' = -\tau_m K_j^T (B_i + \Delta B_i)^T Q_4, \\ \Phi_{59}'' &= -\tau_s K_j^T (B_i + \Delta B_i)^T Q_5, \Phi_{66}'' = -\gamma^2 I, \Phi_{67}'' = -\tau_m (B_i + \Delta B_i)^T Q_3, \Phi_{68}'' = -\tau_m (B_i + \Delta B_i)^T Q_4, \\ \Phi_{69}'' &= -\tau_s (B_i + \Delta B_i)^T Q_5, \Phi_{77}'' = -Q_3, \Phi_{88}'' = -Q_4, \Phi_{99}'' = -Q_5 \end{aligned}$$

According to lemma 4, transforming Φ'' into, $\Phi'' = \Phi''' + \varepsilon M_1 M_1^T + \varepsilon^{-1} E_1^T E_1$ where

$$M_2 = [M^T P \quad 0 \quad 0 \quad 0 \quad 0 \quad 0 \quad \tau_m M^T Q_3 \quad \tau_m M^T Q_4 \quad \tau_s M^T Q_5]^T, E_2 = [N_{ai} \quad N_{bi} K_j \quad 0 \quad 0 \quad -N_{bi} K_j \quad -N_{bi} \quad 0 \quad 0 \quad 0]$$

$$\Phi''' = \begin{bmatrix} \Phi_{11}''' & \Phi_{12}''' & \Phi_{13}''' & \Phi_{14}''' & \Phi_{15}''' & \Phi_{16}''' & \Phi_{17}''' & \Phi_{18}''' & \Phi_{19}''' \\ * & \Phi_{22}''' & \Phi_{23}''' & \Phi_{24}''' & 0 & 0 & \Phi_{27}''' & \Phi_{28}''' & \Phi_{29}''' \\ * & * & \Phi_{33}''' & \Phi_{34}''' & 0 & 0 & 0 & 0 & 0 \\ * & * & * & \Phi_{44}''' & 0 & 0 & 0 & 0 & 0 \\ * & * & * & * & \Phi_{55}''' & 0 & \Phi_{57}''' & \Phi_{58}''' & \Phi_{59}''' \\ * & * & * & * & * & \Phi_{66}''' & \Phi_{67}''' & \Phi_{68}''' & \Phi_{69}''' \\ * & * & * & * & * & * & \Phi_{77}''' & 0 & 0 \\ * & * & * & * & * & * & * & \Phi_{88}''' & 0 \\ * & * & * & * & * & * & * & * & \Phi_{99}''' \end{bmatrix} < 0$$

where * represents the corresponding matrix to be obtained through the symmetric matrix.

$$\begin{aligned} \Phi_{11}''' &= P A_i + A_i^T P + Q_1 - Q_3 - Q_4 + C_i^T C_i, \Phi_{12}''' = P B_i K_j, \Phi_{13}''' = Q_3, \Phi_{14}''' = Q_4, \Phi_{15}''' = -P B_i K_j, \Phi_{16}''' = -P B_i, \\ \Phi_{17}''' &= \tau_m A_i^T Q_3, \Phi_{18}''' = \tau_m A_i^T Q_4, \Phi_{19}''' = \tau_s A_i^T Q_5, \Phi_{22}''' = -2Q_5 + M_{12}^T + M_{12} + \sigma \Xi, \Phi_{23}''' = Q_5 - M_{12}, \Phi_{24}''' = Q_5 - M_{12}^T, \\ \Phi_{27}''' &= \tau_m K_j^T B_i^T Q_3, \Phi_{28}''' = \tau_m K_j^T B_i^T Q_4, \Phi_{29}''' = \tau_s K_j^T B_i^T Q_5, \Phi_{33}''' = Q_2 - Q_1 - Q_3 - Q_5, \Phi_{34}''' = M_{12}^T, \\ \Phi_{44}''' &= -Q_2 - Q_4 - Q_5, \Phi_{55}''' = -\Xi, \Phi_{57}''' = -\tau_m K_j^T B_i^T Q_3, \Phi_{58}''' = -\tau_m K_j^T B_i^T Q_4, \Phi_{59}''' = -\tau_s K_j^T B_i^T Q_5, \Phi_{66}''' = -\gamma^2 I, \\ \Phi_{67}''' &= -\tau_m B_i^T Q_3, \Phi_{68}''' = -\tau_m B_i^T Q_4, \Phi_{69}''' = -\tau_s B_i^T Q_5, \Phi_{77}''' = -Q_3, \Phi_{88}''' = -Q_4, \Phi_{99}''' = -Q_5 \end{aligned}$$

Applying Schur Complement to Φ''' again, we get

$$\Phi'' = \begin{bmatrix} \Phi_{11}''' & \Phi_{12}''' & \Phi_{13}''' & \Phi_{14}''' & \Phi_{15}''' & \Phi_{16}''' & \Phi_{17}''' & \Phi_{18}''' & \Phi_{19}''' & PM & N_{ai}^T \\ * & \Phi_{22}''' & \Phi_{23}''' & \Phi_{24}''' & 0 & 0 & \Phi_{27}''' & \Phi_{28}''' & \Phi_{29}''' & 0 & K_j^T N_{bi}^T \\ * & * & \Phi_{33}''' & \Phi_{34}''' & 0 & 0 & 0 & 0 & 0 & 0 & 0 \\ * & * & * & \Phi_{44}''' & 0 & 0 & 0 & 0 & 0 & 0 & 0 \\ * & * & * & * & \Phi_{55}''' & 0 & \Phi_{57}''' & \Phi_{58}''' & \Phi_{59}''' & 0 & -K_j^T N_{bi}^T \\ * & * & * & * & * & \Phi_{66}''' & \Phi_{67}''' & \Phi_{68}''' & \Phi_{69}''' & 0 & -N_{bi}^T \\ * & * & * & * & * & * & \Phi_{77}''' & 0 & 0 & \tau_m Q_3 M & 0 \\ * & * & * & * & * & * & * & \Phi_{88}''' & 0 & \tau_m Q_4 M & 0 \\ * & * & * & * & * & * & * & * & \Phi_{99}''' & \tau_s Q_5 M & 0 \\ * & * & * & * & * & * & * & * & * & -\varepsilon^{-1} I & 0 \\ * & * & * & * & * & * & * & * & * & * & -\varepsilon I \end{bmatrix} < 0$$

where * represents the corresponding matrix to be obtained through the symmetric matrix.

Before and after multiplying Φ'' with $diag \{P^{-1}, P^{-1}, P^{-1}, P^{-1}, P^{-1}, P^{-1}, Q_3^{-1}, Q_4^{-1}, Q_5^{-1}, I, I\}$

Define $\begin{cases} X = P^{-1}, K_j X = Y_j, \Xi^{-1} = V \\ Q_i^{-1} = R_i (i = 1, 2, \dots, 5) \\ M_{12}^{-1} = R_6 \end{cases}$ and take the following formal transformation:

$$\begin{cases} P^{-1}Q_iP^{-1} \geq 2P^{-1} - Q_i^{-1} = 2X - R_i (i = 1, 2, \dots, 5) \\ P^{-1}\Xi P^{-1} \geq 2P^{-1} - \Xi^{-1} = 2X - V \\ P^{-1}M_{12}P^{-1} \geq 2P^{-1} - M_{12}^{-1} = 2X - R_6 \end{cases}$$

We obtain LMI Φ , where $K_j = Y_j X^{-1}$ is the gain matrix to be designed $\Xi = V^{-1}$, and is the event-triggered weight matrix to be determined.

As previously described, we could make the following conclusions:

$$V(t) + y^T(t)y(t) - \gamma^2 w^T(t)w(t) \leq \sum_{i=1}^r \sum_{j=1}^r h_i h_j \eta^T(t) \Phi \eta(t), \text{ when both } w(t) = 0 \text{ and LMI } \Phi < 0 \text{ are true, then } \dot{V}(t) \leq 0$$

could be driven and the error system is asymptotically stable.

For any non-zero $w(t) \in [0, \infty)$, integral is the previous formula from $t_0 \rightarrow t$ and we obtain

$$V(t) - V(t_0) \leq -\int_{t_0}^t (y^T(t)y(t) - \gamma^2 w^T(t)w(t)) dt$$

Under zero initial condition, when $t \rightarrow \infty$, then $\int_0^\infty y^T(t)y(t) dt \leq \int_0^\infty \gamma^2 w^T(t)w(t) dt$ is tenable, that is, $\|y(t)\|_2 \leq \gamma \|w(t)\|_2$. Thus, the fault model (24) has γ rejection performance of disturbance.

The proof to *theorem 2* is hereby completed.

3.2.2. Design of PFTC

Aimed at system fault model (9), PFTC is designed as follows:

According to formula (7), the state-feedback controller could be written as $u_p(t) = \sum_{i=1}^r h_i(\theta(t)) K_{pi} (x(t - \tau(t)) - e_x(i_k h))$

where K_{pi} represents the gain matrix of the PFTC, which ensures that the system is asymptotically stable in the event of any possible actuator failure.

Theorem 3. Under the event-triggered condition (3) in the DETCS, for the given constants $\sigma \in [0, 1), \tau_m, \tau_M, \tau_s$, if there exist positive definite symmetric matrices $R_i > 0, i = 1, 2, \dots, 5$ and X, V, Y_i for any possible actuator failures in mode set L and acceptable parameter uncertainties, these parameters satisfy the following LMI:

$$\Phi = \begin{bmatrix} \Phi_{11} & \Phi_{12} & \Phi_{13} & \Phi_{14} & \Phi_{15} & \Phi_{16} & \Phi_{17} & \Phi_{18} & \Phi_{19} & \Phi_{110} \\ * & \Phi_{22} & \Phi_{23} & \Phi_{24} & 0 & \Phi_{26} & \Phi_{27} & \Phi_{28} & 0 & \Phi_{210} \\ * & * & \Phi_{33} & \Phi_{34} & 0 & 0 & 0 & 0 & 0 & 0 \\ * & * & * & \Phi_{44} & 0 & 0 & 0 & 0 & 0 & 0 \\ * & * & * & * & \Phi_{55} & \Phi_{56} & \Phi_{57} & \Phi_{58} & 0 & \Phi_{510} \\ * & * & * & * & * & \Phi_{66} & 0 & 0 & \Phi_{69} & 0 \\ * & * & * & * & * & * & \Phi_{77} & 0 & \Phi_{79} & 0 \\ * & * & * & * & * & * & * & \Phi_{88} & \Phi_{89} & 0 \\ * & * & * & * & * & * & * & * & \Phi_{99} & 0 \\ * & * & * & * & * & * & * & * & * & \Phi_{1010} \end{bmatrix} < 0$$

where * represents the corresponding matrix to be obtained through the symmetric matrix.

$$\begin{aligned}
\Phi_{11} &= A_1 X + X A_1^T - 2X - R_1 + R_3 + R_4, \Phi_{12} = B_i L Y_j, \Phi_{13} = 2X - R_3, \Phi_{14} = 2X - R_4, \Phi_{15} = -B_i L Y_j, \Phi_{16} = \tau_m X A_1^T, \\
\Phi_{17} &= \tau_m X A_1^T, \Phi_{18} = \tau_s X A_1^T, \Phi_{19} = M, \Phi_{110} = X N_{ai}^T, \Phi_{22} = \sigma(2X - V) + 2R_5 - R_6 - R_6^T, \Phi_{23} = R_6 - R_5, \\
\Phi_{24} &= R_6^T - R_5, \Phi_{26} = \tau_m Y_j^T L B_i^T, \Phi_{27} = \tau_M Y_j^T L B_i^T, \Phi_{28} = \tau_s Y_j^T L B_i^T, \Phi_{210} = Y_j^T L N_{bi}^T, \Phi_{33} = R_1 + R_3 + R_5 - R_2 - 4X, \\
\Phi_{34} &= 2X - R_6^T, \Phi_{44} = R_2 + R_4 + R_5 - 6X, \Phi_{55} = -2X + V, \Phi_{56} = -\tau_m Y_j^T L B_i^T, \Phi_{57} = -\tau_M Y_j^T L B_i^T, \Phi_{58} = -\tau_s Y_j^T L B_i^T, \\
\Phi_{510} &= -Y_j^T L N_{bi}^T, \Phi_{66} = -R_3, \Phi_{69} = \tau_m M, \Phi_{77} = -R_4, \Phi_{79} = \tau_M M, \Phi_{88} = -R_5, \Phi_{89} = \tau_s M, \Phi_{99} = -\varepsilon^{-1} I, \\
\Phi_{1010} &= -\varepsilon I
\end{aligned}$$

then there exists state-feedback control law (7), which keeps the NNCS fault model (9) asymptotically stable, where controller gain matrix and event-triggered weight matrix could be computed through $k_j = y_j X^{-1}$ and $\Xi = V^{-1}$, respectively.

Remark 3. When $L = \begin{bmatrix} 1 & 0 \\ 0 & 1 \end{bmatrix}$, then $k_j = y_j X^{-1}$ and $\Xi = V^{-1}$ are normal controller gain matrix K_N and event-triggered weight matrix, Ξ_N respectively.

4. SIMULATION AND RESULT ANALYSIS

The NNCS model data in literature [18] and the fuzzy membership function as $M_1(x_2) = \sin^2 x_2$ and $M_2(x_2) = \cos^2 x_2$ are adopted. The system model is expressed as the following T-S fuzzy system of two rules.

$$R^1: \text{ If } x_2 \text{ is } M_1, \text{ then } \begin{cases} x(t) = (A_1 + \Delta A_1)x(t) + (B_1 + \Delta B_1)u(t) + E_1 f(t) \\ y(t) = C_1 x(t) \end{cases}$$

$$R^2: \text{ If } x_2 \text{ is } M_2, \text{ then } \begin{cases} x(t) = (A_2 + \Delta A_2)x(t) + (B_2 + \Delta B_2)u(t) + E_2 f(t) \\ y(t) = C_2 x(t) \end{cases}$$

$$\text{where } A_1 = \begin{bmatrix} -3 & 1 \\ 1 & -1 \end{bmatrix}, A_2 = \begin{bmatrix} -2 & 1 \\ 1 & 0 \end{bmatrix}, B_1 = \begin{bmatrix} 1 & 0 \\ 0 & -0.5 \end{bmatrix}, B_2 = \begin{bmatrix} 1 & 0 \\ 0 & 0.5 \end{bmatrix}, C_1 = \begin{bmatrix} 1 & 0 \\ 0 & 1 \end{bmatrix}, C_2 = \begin{bmatrix} 1 & 0 \\ 0 & -1 \end{bmatrix}$$

Matrices $\Delta A_i, \Delta B_i (i = 1, 2)$ satisfied the following relationship: $[\Delta A_i \quad \Delta B_i] = MF(t)[N_{ai} \quad N_{bi}]$ and

$$N_{ai} = \begin{bmatrix} 0 & 0.2 \\ 0 & 0 \end{bmatrix}, N_{bi} = \begin{bmatrix} 0 & 0.2 \\ 0 & 0 \end{bmatrix}, M = \begin{bmatrix} 0 & 0.1 \\ 0 & 0 \end{bmatrix}, F = \begin{bmatrix} \sin t & 0 \\ 0 & \cos t \end{bmatrix}.$$

We set the parameters as follows: $h = 0.05s, \tau_m = 0.2, \tau_M = 0.3, \tau_s = 0.1, \sigma = 0.3, a = 0.1, b = 0.2, c = 0.5, \gamma = 2.3, \varepsilon = 2$.

$$\text{Actuator fault matrix } L \text{ defined as follows: } L_0 = \begin{bmatrix} 1 & 0 \\ 0 & 1 \end{bmatrix}, L_1 = \begin{bmatrix} 0.2 & 0 \\ 0 & 0.8 \end{bmatrix}, L_2 = \begin{bmatrix} 1 & 0 \\ 0 & 0 \end{bmatrix}.$$

where L_0 means that both of the actuators are normally operated, L_1 means that both of the actuators are at fault to some extent, and L_2 means that one of the actuators is totally faulty and the other one is normally operated.

According to *theorem 3*, we obtain the state-feedback controller gain matrices and the event-triggered weight matrix as the following: $K_1 = \begin{bmatrix} -1.4120 & -2.8050 \\ -0.2423 & -0.4819 \end{bmatrix}, K_2 = \begin{bmatrix} -1.2958 & -2.6347 \\ -0.2858 & -0.5888 \end{bmatrix}, \Xi = \begin{bmatrix} 0.0284 & 0.0490 \\ 0.0490 & 0.0933 \end{bmatrix}.$

Similarly, according to *theorem 1*, we obtain the FDO gain matrices and the residual gain matrix as follows:

$$G_1 = \begin{bmatrix} -0.3101 & 0.6434 \\ 0.4980 & 0.8304 \end{bmatrix}, G_2 = \begin{bmatrix} 0.2178 & 0.6824 \\ 0.5109 & 0.5598 \end{bmatrix}, W = \begin{bmatrix} 1.2910 & -0.1054 \\ -0.1054 & -1.3509 \end{bmatrix}$$

Assuming the system operates normally before $t = 5s$ and known faults begin to appear at the instant $t = 5s$,

unknown fault such as $f = \begin{bmatrix} 0 \\ 0.5 \end{bmatrix}$ begins to appear at time $t = 10s$.

The simulation results are shown as follows: The response curve of the estimation of faults is shown in Fig. (2). Setting initial condition $x(0) = [2-2]^T$, under the aforementioned actuator failures in mode set L , and the results contrasting figures of state component in x_1, x_2 in passive (S1), active (S2), and hybrid (S3) fault-tolerant control are shown in Fig. (3) and Fig. (4), respectively. Similarly, under event-triggered condition (1) in the DETCS and with simulation time, the relationship between release time(s) and release interval of data transmission is shown in Fig. (5).

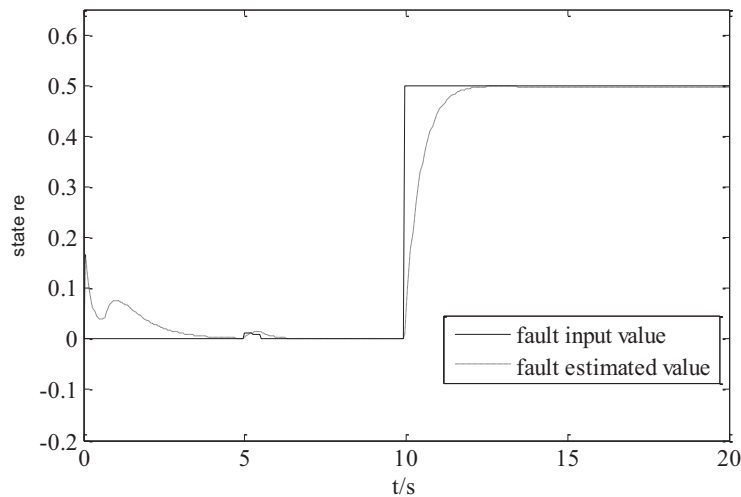


Fig. (2). Response curve of the estimation of faults

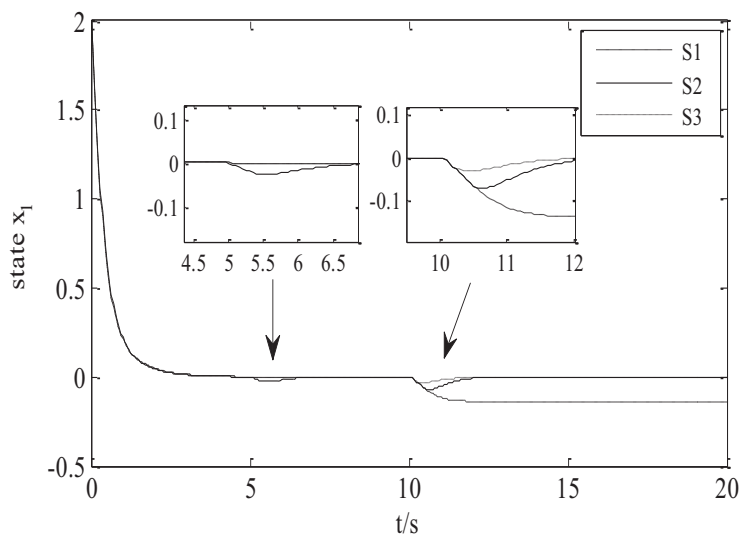


Fig. (3). Response curve of state in different control action.

First, we can draw the conclusion from Fig. (2) that the FDO we designed could estimate the value of any faults timely and accurately compared with fault input value. Although the fault-estimated value lag behind at the instants $t = 5s$ and $t = 10s$, they fit closely at all other times, which illustrated the validity of the fault estimation method that we have adopted.

Figs. (3) and (4) show that when an actuator operates normally, both AFTC and PFTC could maintain system stability; when an actuator appears at known faults ($t = 5s$), PFTC could be tolerant on the fault effectively and keeps the system stable. By contrast, AFTC needs time to reconstruct a controller, according to the detected fault information. This requirement wastes time in helping the system recovery become stable. Thus, a short time of oscillation appeared at first, and then the system performance recovered after compensating the impact of fault to the system; on the contrary, when unknown faults happen ($t = 10s$), PFTC loses the ability to tolerate faults, and the system becomes

unstable. Nonetheless, AFTC begins to show its superiority. AFTC has already estimated the value of fault accurately before the system turned into an unstable state and a reconstructed controller according to the estimated fault value. Therefore, the system became stable again. Adopting hybrid fault-tolerant control method not only can keep the system stable when known failure happens but also can slow down the rate of the system performance deterioration when unknown fault occurs. Meanwhile, reconstructing a controller rapidly makes the system recovery stable. Most of all, hybrid active–passive fault-tolerant control method combined the merits of AFTC and PFTC, which contribute much to the improvement of the system’s performance superiority.

To illustrate that the addition of event generator saves network resources to some extent, we adopt a simulation of 20 s whose horizontal axis represents the instants of data transmission $t_k h$ and vertical axis represents the release interval between $t_k h$ and $i_k h$. Fig. (5) shows that the system sampled 400 times theoretically, but only 128 times transmission practically; the longer the vertical axis value is, the lesser transmission quantity as well as the better quality the system has. Hence, from Fig. (5), we can conclude that the introduction of event-triggered condition reduced network resources and, as a result, increased the efficiency of network utilization.

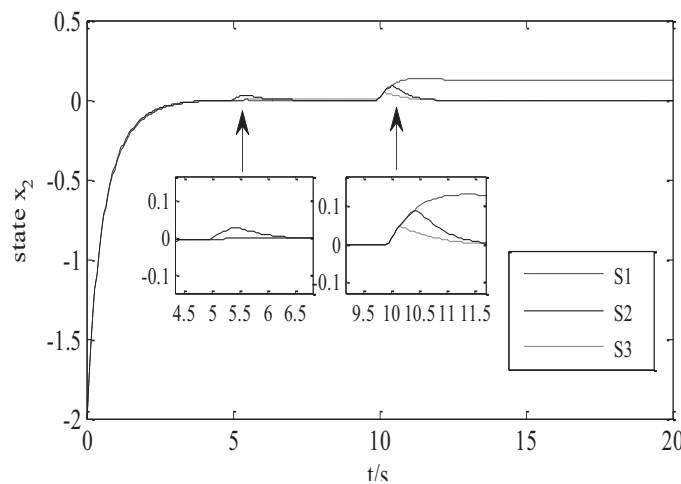


Fig. (4). Response curve of state in different control action.

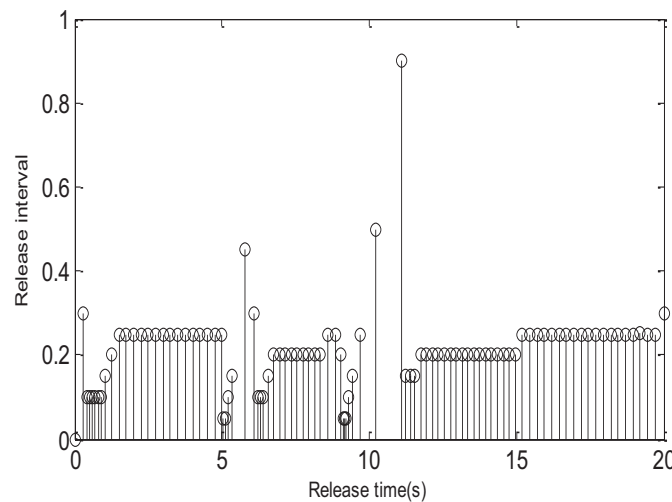


Fig. (5). Release time(s) and release interval of data transmission for hybrid active–passive fault-tolerant control of NNCS based on DETCS.

CONCLUSION

This paper studied uncertain NNCS with time-varying delay and any actuator failure based on DETCS; under the circumstance of reduced network resources, the integrated-design method of FDO and hybrid active–passive robust

fault-tolerant controller is researched. As has been demonstrated in the above chapter, this new method combined the merits of AFTC and PFTC, helped the system in detecting fault information effectively, determined whether the information is time-varying or not, and ensured the stability of the system, whatever type the fault belonged to. Finally, a numerical example is given to show the economy of DETCS to network resources and the effectiveness of the proposed method.

CONFLICT OF INTEREST

The authors confirm that this article content has no conflict of interest.

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